

## LUFTVÄRDIGHETSDIREKTIV (LVD)

A. Flygplan Raytheon (Beech) LVD Nr 2825

Sektion 2. Utlandstillverkad flygmateriel

TITEL:

Ändring av "Limitation Section" i flyghandboken

GÄLLER:

Raytheon, modeller angivna i bifogad kopia av FAA AD 97-25-03

ATGÄRD:

Utför åtgärder angivna i bifogad kopia av FAA AD 97-25-03

TID FÖR ATGÄRD:

Inom 30 dagar räknat från 21 januari 1998

**UNDERLAG:** 

FAA AD 97-25-03

**REFERENS:** 

FAA AD 97-25-03

**BESLUTSDATUM:** 

1998-01-15

LFS 1998:3

Åtgärder enligt LVD utgör nödvändig förutsättning för ifrågavarande flygmateriels luftvärdighet. Referens BCL M 1.11.

Anteckning om åtgärd, som vidtagits i enlighet med LVD, skall införas i teknisk journal för berörd flygmateriel med hänvisning till ifrågavarande LVD-nummer. Angivet underlag refererar till senast gällande revision/utgåva. LVD utges i luftfartsverkets författningssamlingar LFS.

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## **AIRWORTHINESS DIRECTIVE**

REGULATORY SUPPORT DIVISION P.O. BOX 26460 OKLAHOMA CITY, OKLAHOMA 73125-0460

U.S. Department of Transportation Federal Aviation Administration

The following Ainvorthness Directive issued by the Federal Aviation Administration in accordance with the provisions of Federal Aviation Regulations, Part 39, applies to an aircraft model of which our records indicate you may be the registered owner. Ainvorthness Directives affect aviation safety and are regulations which require immediate attention. You are cautioned that no person may operate an aircraft to which an Ainvorthness Directive applies, except in accordance with the requirements of the Ainvorthness Directive (reference FAR Subpart 39.3).

97-25-03 RAYTHEON AIRCRAFT COMPANY (formerly Beech Aircraft Corporation): Amendment 39-10226; Docket No. 97-CE-20-AD.

Applicability: Models 65-90, 65-A90, 65-A90-1, 65-A90-3, 65-A90-4, B90, C90, C90(SE), C90A, C90B, E90, F90, H90, 99, 99A, A99, A99A, B99, C99, 100, A100, A100A, A100C, B100, 200, 200C, 200CT, 200T, A200, A200CT, B200, B200C, B200CT, B200CT, 300, B300C, 1900, 1900C, 1900D, and 2000 airplanes, all serial numbers, certificated in any category.

NOTE 1: This AD applies to each airplane identified in the preceding applicability provision, regardless of whether it has been modified, altered, or repaired in the area subject to the requirements of this AD. For airplanes that have been modified, altered, or repaired so that the performance of the requirements of this AD is affected, the owner/operator must request approval for an alternative method of compliance in accordance with paragraph (e) of this AD. The request should include an assessment of the effect of the modification, alteration, or repair on the unsafe condition addressed by this AD; and, if the unsafe condition has not been eliminated, the request should include specific proposed actions to address it.

Compliance: Required within the next 30 days after the effective date of this AD, unless already accomplished.

To prevent nose down pitch and a descent rate leading to aircraft damage and injury to personnel caused by the power levers being positioned below the flight idle stop or the power levers being lifted while the airplane is in flight, accomplish the following:

(a) Amend the Limitations Section of the airplane flight manual (AFM) by inserting the following language:

"Do not lift the power levers in flight. Lifting the power levers in flight or moving the power levers in flight below the flight idle position could result in nose down puch and a descent rate leading to aircraft damage and injury to personnel."

- (b) This action may be accomplished by incorporating a copy of this AD into the Limitations Section of the AFM.
- (c) Amending the AFM, as required by this AD, may be performed by the owner/operator holding at least a private pilot certificate as authorized by section 43.7 of the Federal Aviation Regulations (14 CFR 43.7), and must be entered into the aircraft records showing compliance with this AD in accordance with section 43.9 of the Federal Aviation Regulations (14 CFR 43.9).
- (d) Special flight permits may be issued in accordance with sections 21.197 and 21.199 of the Federal Aviation Regulations (14 CFR 21.197 and 21.199) to operate the airplane to a location where the requirements of this AD can be accomplished.
- (e) An alternative method of compliance or adjustment of the compliance time that provides an equivalent level of safety may be approved by the Manager, Wichita Aircraft Certification Office (ACO), FAA, 1801 Airport Road, Wichita, Kansas. The request shall be forwarded through an appropriate FAA Maintenance Inspector, who may add comments and then send it to the Manager, Wichita ACO.

NOTE 2: Information concerning the existence of approved alternative methods of compliance with this AD, if any, may be obtained from the Wichita ACO.

- (f) Information related to this AD may be examined at the FAA, Central Region, Office of the Regional Counsel, Room 1558, 601 E. 12th Street, Kansas City, Missouri 64106.
  - (g) This amendment (39-10226) becomes effective on January 21, 1998.

## FOR FURTHER INFORMATION CONTACT:

William Schinstock, Aerospace Engineer, Wichita Aircraft Certification Office, FAA, 1801 Airport Road, Wichita, Kansas 67209; telephone (316) 946-4162; facsimile (316) 946-4407.