
Sektion 2. Utlandstillverkad flygmateriel

TITEL: Kontroll av rotorkopplingens fjäderenhet

GÄLLER: Modellerna 269 A, 269 A-1, 269 B, 269 C och TH-55 A med installerad fjäderhållare av aluminium P/N 269 A 5452, 269 A 5452-3, 269 A 5452-5 eller 269 A 5483-7.

ÅTGÄRD: Utför åtgärder angivna i Schweizer Service Bulletin 256.2 daterad 11 juni 1993 eller senare utgåva.

TID FÖR ÅTGÄRD: Inom 5 flygtimmar och därefter i intervall av 100 flygtimmar.

UNDERLAG: Schweizer Service Bulletin 256.2 daterad 11 juni 1993 eller senare utgåva.

REFERENS: FAA AD 95-03-12 (kopia bifogad).

UTGIVNINGS-DATUM: 1995-04-06

LFS: 1995:20

Åtgärd enligt LVD utgör nödvändig förutsättning för ifrågavarande flygmateriels luftvärdighet. Referens BCL M 1.11. Anteckning om åtgärd, som vidtagits i enlighet med LVD, skall införas i teknisk journal för berörd flygmateriel med hänvisning till ifrågavarande LVD-nummer. Angivet underlag refererar till senaste gällande revision/utgåva. LVD utges i luftfartsverkets författningssamlingar LFS.



AIRWORTHINESS DIRECTIVE

FLIGHT STANDARDS SERVICE
REGULATORY SUPPORT DIVISION
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U.S. Department
of Transportation
Federal Aviation
Administration

The following Airworthiness Directive issued by the Federal Aviation Administration in accordance with the provisions of Federal Aviation Regulations, Part 39, applies to an aircraft model of which our records indicate you may be the registered owner. Airworthiness Directives affect aviation safety and are regulations which require immediate attention. You are cautioned that no person may operate an aircraft to which an Airworthiness Directive applies, except in accordance with the requirements of the Airworthiness Directive (reference FAR Subpart 39.3).

95-03-12 SCHWEIZER AIRCRAFT CORPORATION AND HUGHES HELICOPTERS, INC.:
Amendment 39-9148. Docket No. 94-SW-15-AD. Supersedes Priority Letter AD 93-03-01, issued on March 4, 1993.

Applicability: Model 269A, 269A-1, 269B, 269C, and TH-55A series helicopters, with aluminum spring retainer, part number (P/N) 269A5452, P/N 269A5452-3, P/N 269A5452-5, or P/N 269A5483-7, installed, certificated in any category.

Compliance: Required as indicated, unless accomplished previously.

To prevent failure of the aluminum spring retainer, loss of power to the rotor drive system, and a subsequent forced landing, accomplish the following:

(a) Within the next 5 hours time-in-service after the effective date of this AD, and thereafter at intervals not to exceed 100 hours time-in-service from the last inspection, visually inspect the clutch control spring assembly for component wear in accordance with the provisions of Part I, paragraph a(2) of Schweizer Service Bulletin (SB) B-256.2, dated June 11, 1993.

(b) If worn parts are found during the inspections accomplished in accordance with paragraph (a) of this AD, before the next flight, disassemble and inspect the clutch control spring assembly and replace parts found to be unairworthy with airworthy parts in accordance with Part I, paragraph b. of SB B-256.2, dated June 11, 1993.

(c) An alternative method of compliance or adjustment of the compliance time that provides an acceptable level of safety may be used when approved by the Manager, New York Aircraft Certification Office, FAA. Operators shall submit their requests through an FAA Principal Maintenance Inspector, who may concur or comment and then send it to the Manager, New York Aircraft Certification Office.

NOTE: Information concerning the existence of approved alternative methods of compliance with this AD, if any, may be obtained from the New York Aircraft Certification Office.

(d) Special flight permits may be issued in accordance with sections 21.197 and 21.199 of the Federal Aviation Regulations (14 CFR 21.197 and 21.199) to operate the helicopter to a location where the inspection requirements of this AD can be accomplished.

(e) The inspections and replacement, if necessary, shall be done in accordance with SB B-256.2, dated June 11, 1993. This incorporation by reference was approved previously by the Director of the Federal Register in accordance with 5 U.S.C. 552(a) and 1 CFR part 51 as of September 1, 1994 (59 FR 38354, July 28, 1994). Copies may be obtained from Schweizer Aircraft Corporation, P.O. Box 147, Elmira, New York 14902. Copies may be inspected at the FAA, Office of the Assistant Chief Counsel, 2601 Meacham Blvd., Room 663, Fort Worth, Texas; or at the Office of the Federal Register, 800 North Capitol Street, NW., suite 700, Washington, DC.

(f) This amendment becomes effective on March 1, 1995.

FOR FURTHER INFORMATION CONTACT:

Mr. Raymond Reinhardt, Aerospace Engineer, New York Aircraft Certification Office, FAA, New England Region, 10 Fifth Street, Valley Stream, New York 11581, telephone (516) 256-7532; fax (516) 568-2716.