
Sektion 2. Utlandstillverkad flygmateriel

TITEL: Sprickkontroll av fästen för landställets stöttor

GÄLLER: Socata TB20 och TB21 S/N 1 t o m 9999

ÅTGÄRD: Utför åtgärder angivna i bifogad kopia av DGAC AD 1994-266(A) R3

TID FÖR ÅTGÄRD: Inom tider och intervaller angivna i DGAC AD 1994-266(A) R3

UNDERLAG: DGAC AD 1994-266(A) R3
SOCATA TB Aircraft SB 10.080.57 Rev 3

REFERENS: DGAC AD 1994-266(A) R3

BESLUTSDATUM: 1999-03-11

LFS 1999:41

Åtgärder enligt LVD utgör nödvändig förutsättning för ifrågavarande flygmateriels luftvärdighet. Referens BCL M 1.11.

Anteckning om åtgärd, som vidtagits i enlighet med LVD, skall införas i teknisk journal för berörd flygmateriel med hänvisning till ifrågavarande LVD-nummer. Angivet underlag refererar till senast gällande revision/utgåva. LVD utges i luftfartsverkets författningssamlingar LFS.

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GSAC

AIRWORTHINESS DIRECTIVE

released by DIRECTION GENERALE DE L'AVIATION CIVILE

Inspection and/or modifications described below are mandatory. No person may operate a product to which this Airworthiness Directive applies except in accordance with the requirements of this Airworthiness Directive.

Translation of 'Consigne de Navigabilité' ref. : 1994-266(A) R3
In case of any difficulty, reference should be made to the French original issue.

SOCATA

TB20, TB21 Airplanes

Main landing gear

This Airworthiness Directive concerns TB20 and TB21 aircraft S/N 1 to 9999.

In order to ensure the integrity of the hinged strut attachment bearings by a crack detection inspection, the following measures are made mandatory from the effective date of this Airworthiness Directive.

A) Attachment bearing that has reached or exceeded 6000 landings or 4000 flight hours

Inspect, within next 100 hours T.I.S. or at the latest on October 31, 1995, the hinged strut attachment bearings of main landing gear that have reached or exceeded 6000 landings or 4000 flight hours (whichever occurs first). Then repeat this inspection every 1500 landings or every 1000 flight hours (whichever occurs first).

B) Attachment bearing that has not reached 6000 landings or 4000 flight hours

Inspect the hinged strut attachment bearings of main landing gear when aircraft reaches 6000 landings or 4000 flight hours (whichever occurs first). Then repeat this inspection every 1500 landings or every 1000 flight hours (whichever occurs first).

If cracks are detected, replace the bearing in conformity with instructions given in SOCATA TB aircraft Service Bulletin No. 10-080-57 Rev.3 and recommence the cycle of inspections at 6000 landings or 4000 flight hours (whichever occurs first).

Record the application of this Airworthiness Directive in the aircraft Log book.

Ref : SOCATA TB aircraft SB 10-080-57 Rev.3

This Revision 3 replaces AD 1994-266(A) R2 dated December 06, 1995.

EFFECTIVE DATE : DECEMBER 17, 1994

(Same as the original AD)

n/JB

February 10, 1999

SOCATA
TB20, TB21 Airplanes

1994-266(A) R3