

Sektion 2. Utlandstillverkad flygmateriel

TITEL: Inspektion av motordriven bränslepumps dräneringsrör

GÄLLER: Modell 269 helikopter med bränsleinsprutningsmotorer.
Alla modeller 269A-1 och 269B och modell 269C serie nr 0004-1480 och 1482-1486.

ÅTGÄRD: Utför åtgärder i enlighet med bifogad kopia av FAA AD 93-05-11.

**TID FÖR
ÅTGÄRD:** Inom 5 flygtimmar räknat från detta LVD's utgivningsdatum.

UNDERLAG: FAA AD 93-05-11.
Schweizer Aircraft Corporation SB B-235.1 daterad 20 mars 1992.

REFERENS: FAA AD 93-05-11.

**UTGIVNINGS-
DATUM:** 1993-06-03

LFS: 1993:20

Åtgärd enligt LVD utgör nödvändig förutsättning för ifrågavarande flygmateriels luftvärdighet. Referens BCL M 1.11.
Anteckning om åtgärd, som vidtagits i enlighet med LVD, skall införas i teknisk journal för berörd flygmateriel med hänvisning till ifrågavarande LVD-nummer. Angivet underlag refererar till senaste gällande revision/utgåva. LVD utges i luftfartsverkets författningssamlingar LFS.



AIRWORTHINESS DIRECTIVE

FLIGHT STANDARDS SERVICE
REGULATORY SUPPORT DIVISION
P.O. BOX 26460
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U.S. Department
of Transportation
**Federal Aviation
Administration**

The following Airworthiness Directive issued by the Federal Aviation Administration in accordance with the provisions of Federal Aviation Regulations, Part 39, applies to an aircraft model of which our records indicate you may be the registered owner. Airworthiness Directives affect aviation safety and are regulations which require immediate attention. You are cautioned that no person may operate an aircraft to which an Airworthiness Directive applies, except in accordance with the requirements of the Airworthiness Directive (reference FAR Subpart 39.3).

93-05-11 SCHWEIZER AIRCRAFT CORPORATION AND HUGHES HELICOPTERS, INC.: Amendment 39-8515. Docket No. 92-ASW-47.

Applicability: Model 269A helicopters with fuel injected engines; all Model 269A-1 and 269B helicopters; and Model 269C helicopters, serial numbers (S/N) 0004 through 1480 and 1482 through 1486; equipped with an engine-driven fuel pump drain tube assembly (tube assembly), part number (P/N) 269A8326-95, certificated in any category.

Compliance: Required as indicated, unless accomplished previously.

To prevent a restriction to the movement of the engine throttle control linkage that could result in loss of control of engine power due to tube assembly interference, accomplish the following:

(a) Within 5 hours time in service after the effective date of this AD, perform a one-time inspection of the length of the tube assembly, P/N 269A8326-95, in accordance with Part I of Schweizer Aircraft Corporation (Schweizer) Service Bulletin B-235.1, dated March 20, 1992 (SSB). If the tube assembly length is less than 7.50 inches, before next flight, install tube assembly, P/N 269A8326-235, with a clamp, P/N MS21333-98, in accordance with Part III of the SSB.

(b) Within 5 hours time in service after the effective date of this AD, and thereafter at intervals not to exceed 25 hours time in service from the last check, check the tube assembly, P/N 269A8326-95, for security and clearance from the throttle linkage as follows:

(1) Move the throttle linkage from fully open to fully closed and verify that the tube assembly is not loose and does not interfere with movement of the throttle linkage.

(2) If the tube assembly interferes with movement of the throttle linkage, before next flight--

(i) Install tube assembly, P/N 269A8326-235, with a clamp, P/N MS21333-98, in accordance with Part III of the SSB; or

(ii) Secure tube assembly, P/N 269A8326-95, in accordance with Part II b of the SSB and conduct repetitive checks as prescribed by this paragraph.

(3) After the initial check, the repetitive check contained in this paragraph may be performed by the pilot and must be recorded in accordance with FAR 43.9.

(c) Within 6 months after the effective date of this AD, replace the tube assembly, P/N 269A8326-95, with an improved tube assembly, P/N 269A8326-235, and clamp, P/N MS21333-98, in accordance with Part III of the SSB.

(d) Installation of the improved tube assembly, P/N 269A8326-235, and clamp, P/N MS21333-98, constitutes terminating action for the requirements of this AD.

(e) An alternative method of compliance or adjustment of the compliance time, which provides an acceptable level of safety, may be used when approved by the Manager, New York Aircraft Certification Office, 181 South Franklin Avenue, Room 202, Valley Stream, New York 11581. Operators shall submit their requests through an FAA Principal Maintenance Inspector, who may concur or comment and then send it to the Manager, New York Aircraft Certification Office.

NOTE: Information concerning the existence of approved alternative methods of compliance with this AD, if any, may be obtained from the Manager, New York Aircraft Certification Office.

(f) Special flight permits may be issued in accordance with FAR 21.197 and 21.199 to operate the helicopter to a location where the requirements of this AD can be accomplished.

(g) The inspections and modifications shall be done in accordance with Schweizer Service Bulletin B-235.1, dated March 20, 1992. This incorporation by reference was approved by the Director of the Federal Register in accordance with 5 U.S.C. 552(a) and 1 CFR Part 51. Copies may be obtained from Schweizer Aircraft Corporation, P.O. Box 147, Elmira, New York 14902. Copies may be inspected at the FAA, Office of the Assistant Chief Counsel, 4400 Blue Mound Road, Bldg. 3B, Room 158, Fort Worth, Texas; or at the Office of the Federal Register, 800 North Capitol Street, NW., suite 700, Washington, DC.