
Sektion 2. Utlandstillverkad flygmateriel

TITEL: Korrosionskontroll av sidorodrets balansvikt

GÄLLER: TB20 och TB21, S/N angivna i bifogad kopia av DGAC AD 93-012(A)R2

ÅTGÄRD: Utför åtgärder angivna i SOCATA TB Service Bulletin Nr TB 10-071-55 Rev. 3 eller senare utgåva

TID FÖR ÅTGÄRD: Enligt DGAC AD 93-012(A)R2 eller senare utgåva

UNDERLAG: SOCATA TB Airplanes Service Bulletin Nr TB 10-071-55 Rev. 3 eller senare utgåva

REFERENS: DGAC AD 93-012(A)R2

BESLUTSDATUM: 1998-04-07

LFS 1998:21

Åtgärder enligt LVD utgör nödvändig förutsättning för ifrågavarande flygmateriels luftvärdighet. Referens BCL M 1.11.

Anteckning om åtgärd, som vidtagits i enlighet med LVD, skall införas i teknisk journal för berörd flygmateriel med hänvisning till ifrågavarande LVD-nummer. Angivet underlag refererar till senast gällande revision/utgåva. LVD utges i luftfartsverkets författningssamlingar LFS.

Postadress	Gatuadress	Telefonnummer	Telegram	Telex
601 79 NORRKÖPING	Vikboplan 11	011-192000	Civilair Norrköping	62450

AIRWORTHINESS DIRECTIVE

released by DIRECTION GENERALE DE L'AVIATION CIVILE

Inspection and/or modifications described below are mandatory. No person may operate a product to which this

Airworthiness Directive applies except in accordance with the requirements of this Airworthiness Directive

Translation of 'Consigne de Navigabilité'
ref.: 93-012(A)R2

In case of any difficulty, reference should
be made to the French original issue.

SOCATA

TB 20 and TB 21 Airplanes

Rudder balance weight

This Airworthiness Directive concerns SOCATA TB 20 and TB 21 airplanes from S/N 1 to S/N 1554, S/N 1754 to S/N 1757 and S/N 1767 to S/N 1821 inclusive.

In order to ensure integrity of balance weight/stiffeners junction, following measures are made mandatory from the effective date of this Airworthiness Directive.

- Not later than June 30, 1998 for S/N 1 to S/N 1554 and December 31, 1998 for S/N 1754 to S/N 1757 and S/N 1767 to S/N 1821, remove the balance weight and detect all traces of galvanic corrosion, in compliance with instructions described in SOCATA TB Airplanes Service Bulletin N° 10-071-55 Rev. 3.

A - If no corrosion trace is detected, reinstall the balance weight.

B - If slight corrosion traces are detected, apply the instructions and treatment described in Chapter II.16 of the Manufacturer Maintenance Manual.

C - If corrosion is too important, replace one or both stiffeners.

- In any case, MASTINOX 6856 K will be interposed during installation of the balance weight.

Record the application of this Airworthiness Directive on the aircraft log book.

Ref : SOCATA TB Airplanes Service Bulletin TB 10-071-55 Rev. 3

.../...

n/JYR

March 11, 1998

SOCATA
TB 20 and TB 21 Airplanes

93-012(A)R2

The present Revision 2 replaces AD 93-012(A)R1 dated March 30, 1994.

EFFECTIVE DATES :

Original AD : FEBRUARY 27, 1993

Revision 2 : MARCH 21, 1998