

Sektion 2. Utlandstillverkad flygmateriel

TITEL: Färgmärkning av hävarm / styrlina i skevroderssystemet

GÄLLER: PA-31 alla modeller alla S/N.

ÅTGÄRD: Utför färgmärkning i enlighet med figur 1 i bifogad kopia av FAA AD 92-27-05.

TID FÖR ÅTGÄRD: Vid demontering av berörd hävarm / styrlina men senast inom 12 månader räknat från detta LVD's utgivningsdatum.

UNDERLAG: FAA AD 92-27-05

REFERENS: FAA AD 92-27-05

UTGIVNINGSDATUM: 1993-01-28

LFS: 1993:3

Åtgärd enligt LVD utgör nödvändig förutsättning för ifrågavarande flygmateriels luftvärdighet. Referens BCL M 1.11. Anteckning om åtgärd, som vidtagits i enlighet med LVD, skall införas i teknisk journal för berörd flygmateriel med hänvisning till ifrågavarande LVD-nummer. Angivet underlag refererar till senaste gällande revision/utgåva. LVD utges i luftfartsverkets författningssamlingar LFS.



AIRWORTHINESS DIRECTIVE

FLIGHT STANDARDS SERVICE
REGULATORY SUPPORT DIVISION
P.O. BOX 26460
OKLAHOMA CITY, OKLAHOMA 73125-0460

U.S. Department
of Transportation
**Federal Aviation
Administration**

The following Airworthiness Directive issued by the Federal Aviation Administration in accordance with the provisions of Federal Aviation Regulations, Part 39, applies to an aircraft model of which our records indicate you may be the registered owner. Airworthiness Directives affect aviation safety and are regulations which require immediate attention. You are cautioned that no person may operate an aircraft to which an Airworthiness Directive applies, except in accordance with the requirements of the Airworthiness Directive (reference FAR Subpart 39.3).

92-27-05 PIPER AIRCRAFT CORPORATION: Amendment 39-8438. Docket No. 91-CE-14-AD.

Applicability: PA-31 Series airplanes (all models and serial numbers), certificated in any category.

Compliance: Required the next time the aileron cables are disconnected for any reason, but no later than 12 calendar months after the effective date of this AD, unless already accomplished.

To prevent improper connection of the aileron control systems, which could result in loss of lateral control of the airplane, accomplish the following:

(a) Using red enamel paint, color code the aileron balance cable and the bellcrank lug of both aileron control systems in accordance with Figure 1 of this AD.

(b) Special flight permits may be issued in accordance with FAR 21.197 and 21.199 to operate airplanes to a location where the requirements of this AD can be accomplished.

(c) An alternative method of compliance or adjustment of the compliance time that provides an equivalent level of safety may be approved by the Manager, Atlanta Aircraft Certification Office, 1669 Phoenix Parkway, Suite 210C, Atlanta, Georgia 30349. The request shall be forwarded through an appropriate FAA Maintenance Inspector, who may add comments and then send it to the Manager, Atlanta Aircraft Certification Office.

NOTE: Information concerning the existence of approved alternative methods of compliance with this AD, if any, may be obtained from the Atlanta Aircraft Certification Office.

(d) All persons affected by this directive may examine any information that is applicable to this AD at the FAA, Central Region, Office of the Assistant Chief Counsel, Room 1558, 601 E. 12th Street, Kansas City, Missouri 64106.

(e) This amendment becomes effective on January 29, 1993.

FOR FURTHER INFORMATION CONTACT:

Mr. Charles Perry, Aerospace Engineer, FAA, Atlanta Aircraft Certification Office, 1669 Phoenix Parkway, Suite 210C, Atlanta, Georgia 30349; Telephone (404) 991-2910; Facsimile (404) 991-3606.

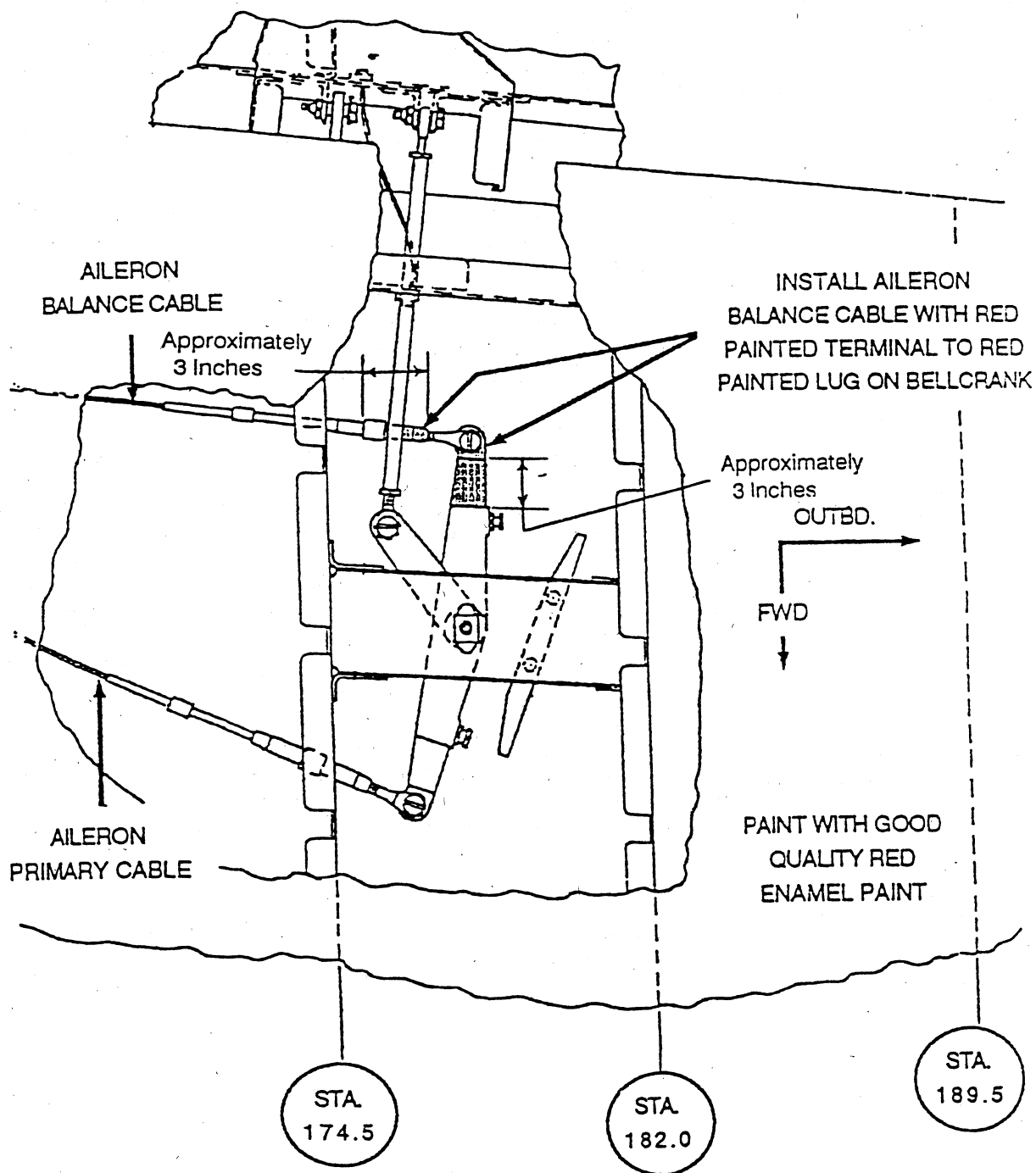


FIGURE 1
VIEW LOOKING THRU TOP WING SKIN (LEFT WING)
RIGHT WING OPPOSITE