

Sektion 2. Utlandstillverkad flygmateriel

TITEL: Kontroll av stjärtrotorns bladbalk

GÄLLER: AS 355 med installerad stjärtrotor PN 350A12.0030, 355A12.031, 355A12.040, 355A12.0050.

ÅTGÄRD: Utför åtgärder angivna i bifogad kopia av DGAC AD 84-045-022(B)R3 och EUROCOPTER FRANCE AS 355 Service Bulletin nr 05-06 Rev. 5.

TID FÖR ÅTGÄRD: Inom 30 flygtimmar och därefter vid intervaller av 400 flygtimmar, samt i DGAC AD 84-045-22(B)R3 "Note:" på sid 2 angivna villkor räknat från detta LVD's utgivningsdatum.

UNDERLAG: DGAC AD 84-045-22(B)R3
EUROCOPTER FRANCE AS355 Service Bulletin nr 05-06 Rev. 5.

REFERENS: DGAC AD 84-045-22(B)R3

UTGIVNINGS-DATUM: 1993-01-28

LFS: 1993:3

Åtgärd enligt LVD utgör nödvändig förutsättning för ifrågavarande flygmateriels luftvärdighet. Referens BCL M 1.11. Anteckning om åtgärd, som vidtagits i enlighet med LVD, skall införas i teknisk journal för berörd flygmateriel med hänvisning till ifrågavarande LVD-nummer. Angivet underlag refererar till senaste gällande revision/utgåva. LVD utges i luftfartsverkets författningssamlingar LFS.

BUREAU VERITAS

AIRWORTHINESS DIRECTIVE

released by DIRECTION GENERALE DE L'AVIATION CIVILE

Inspections and/or modifications described below are mandatory. No person may operate a product to which this Airworthiness Directive applies except in accordance with the requirements of this Airworthiness Directive

Translation of 'Consigne de Navigabilité'

Réf.: 84-045-022(B)R3

In case of any difficulty, reference should be made to the French original issue.

EUROCOPTER FRANCE**AS 355 Helicopters**

Tail rotor blade spars

The present Airworthiness Directive concerns to AS 355 helicopters, all version fitted with tail rotor blades assemblies P/N :

- | | | |
|---------------|---|----------|
| - 350A12.0030 |) | |
| - 355A12.0031 | (| all |
| - 355A12.0040 |) | versions |
| - 355A12.0050 | (| |

Following the discovery in service of several cases of cracks at the blade root on the spar of tail rotor blade assemblies possibly resulting in a severe unbalance, the following measures have been rendered mandatory :

- 1/ From the effective date of the present Airworthiness Directive and at intervals not exceeding 30 flight hours, without removing any components, check for abnormal noise at the root of each tail rotor blade according to the instructions given in paragraph 1C.1 of EUROCOPTER FRANCE AS 355 Service Bulletin N° 05-06 Revision 5.

NOTE :

If abnormal noise is heard, proceed, before any flight, with the check with component removal in accordance with paragraph 1C.2 of the referenced Service Bulletin.

.../...

December 23, 1992

EUROCOPTER FRANCE
AS 355 Helicopters

84-045-022(B)R3

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- 2/ From the effective date of the present Directive and at intervals not exceeding 400 flight hours, with component removal, check for cracks or splinter in the areas bordering the tail rotor blade spars according to the instructions given in paragraph 1C.2 of EUROCOPTER FRANCE AS 355 Service Bulletin N° 05-06, Revision 5.

NOTE :

This check is also to be performed each time the laminated half-bearings are replaced and the event of balancing difficulties or an incident which could adversely affect the tail rotor. If it is carried out when replacing the laminated half-bearings, less than 50 hours before reaching the 400-hour limit indicated in paragraph 2/ above, it is not required on this occasion.

Ref. : EUROCOPTER FRANCE AS 355 Service Bulletin N°05-06 Rev.5

The present Revision 3 replaces the Airworthiness Directive 84-045-022(B)R2 dated September 05, 1990.

EFFECTIVE DATES :

Original AD and Rev. 1 & 2 : APRIL 4, 1984
Revision 3 : JANUARY 02, 1993