

Sektion 2. Utlandstillverkad flygmateriel

<b><u>TITEL:</u></b>	<b>Märkning av höjdrodrens trimdomkrafter.</b>
<b><u>GÄLLER:</u></b>	Beech modellerna 33, 34, 35, 36, 45, 55, 56, 58 och 95 med S/N enligt bifogad kopia av FAA AD 91-17-01.
<b><u>ÅTGÄRD:</u></b>	Utför märkning av trimdomkrafterna i enlighet med Beech Service Bulletin nr 2399 daterad mars 1991 eller senare utgåva.
<b><u>TID FÖR ÅTGÄRD:</u></b>	Nästa gång trimdomkraften är demonterad av någon orsak men ej senare än 12 månader räknat från detta LVD's utgivningsdatum.
<b><u>UNDERLAG:</u></b>	FAA AD 91-17-01 Beech Service Bulletin nr 2399 daterad mars 1991 eller senare utgåva.
<b><u>REFERENS:</u></b>	FAA AD 91-17-01.
<b><u>UTGIVNINGS- DATUM:</u></b>	1991-12-12

**LFS: 1991:32**

Åtgärd enligt LVD utgör nödvändig förutsättning för ifrågavarande flygmateriels luftvärdighet. Referens BCL M 1.11. Anteckning om åtgärd, som vidtagits i enlighet med LVD, skall införas i teknisk journal för berörd flygmateriel med hänvisning till ifrågavarande LVD-nummer. Angivet underlag refererar till senaste gällande revision/utgåva. LVD utges i luftfartsverkets författningssamlingar LFS.



U.S. Department  
of Transportation  
**Federal Aviation  
Administration**

## AIRWORTHINESS DIRECTIVE

AVIATION STANDARDS NATIONAL FIELD OFFICE  
P.O. BOX 26460  
OKLAHOMA CITY, OKLAHOMA 73125

The following Airworthiness Directive issued by the Federal Aviation Administration in accordance with the provisions of Federal Aviation Regulations, Part 39, applies to an aircraft model of which our records indicate you may be the registered owner. Airworthiness Directives affect aviation safety. They are regulations which require immediate attention. You are cautioned that no person may operate an aircraft to which an Airworthiness Directive applies, except in accordance with the requirements of the Airworthiness Directive (FAR 39.3).

**91-17-01 BEECH:** Amendment 39-7099. Docket No. 90-CE-71-AD.

**Applicability:** The following Model airplanes, certificated in any category:

### Models

35-33, 35-A33, 35-B33,  
35-C33, E33, F33, G33

35-C33A, E33A, F33A

E33C, F33C

36, A36

A36TC, B36TC

T34C-1

34C

T-34C

45

A45

### Serial Numbers

CD-1 through CD-981 and  
CD-983 through CD-1304

CE-1 through CE-235,  
CE-249, CE-250, CE-256,  
CE-260, CE-264 through  
CE-268, and CE-270  
through CE-1565

CJ-1 through CJ-179

E-1 through E-2103,  
E-2105 through E-2110

EA-1 through EA-319, and  
EA-321 through EA-388

GM-1 through GM-142

GP-1 through GP-50

GL-1 through GL-353

G-3 through G-6

G-7 through G-156,  
G-257 through G-306,  
G-696 through G-845,  
CG-1 through CG-47,  
CG-58 through CG-60,  
CG-68, CG-73, CG-75,  
CG-78, CG-79,  
CG-105, CG-106,  
CG-108, CG-111  
through CG-179,  
CG-200 through  
CG-223, CG-279  
through CG-319

<u>Models</u>	<u>Serial Numbers</u>
D45	BG-1 through BG-423
95	TD-2 through TD-302
B95	TD-303 through TD-452
B95A	TD-453 through TD-533
D95A	TD-534 through TD-707
E95	TD-708 through TD-721
95-55	TC-1 through TC-190
95-A55	TC-191 through TC-349, TC-351 through TC-370, and TC-372 through TC-501
95-B55, 95-B55A	TC-371, TC-502 through TC-2456
95-C55	TC-350
95-C55A	TE-1 through TE-49, and TE-51 through TE-451
D55, D55A	TE-452 through TE-767
E55, E55A	TE-768 through TE-1201
56TC	TG-2 through TG-83
A56TC	TG-84 through TG-94
58, 58A	TH-1 through TH-1388, and TH-1390 through TH-1395
58P, 58PA	TJ-3 through TJ-435, and TJ-437 through TJ-443
58TC, 58TCA	TK-1 through TK-150

Compliance: Required the next time the elevator trim tab actuators are removed for any reason, but no later than 12 calendar months after the effective date of this AD, unless already accomplished.

To prevent loss of control of the airplane because of interchanging the right-hand and left-hand elevator trim tab actuators, accomplish the following:

(a) Paint a stripe on each stabilizer rear spar (right-hand black; and left-hand blue) in accordance with the Accomplishment Instructions of Beech Service Bulletin No. 2399, dated March 1991.

(b) Remove the cover over the actuator inspection hole on each stabilizer and paint the inspection hole ledges (right-hand black; and left-hand blue) in accordance with the Accomplishment Instructions of Beech Service Bulletin No. 2399, dated March 1991.

(c) Paint a stripe .50 by 1 inch on each actuator housing through the inspection holes (right-hand black; and left-hand blue) in accordance with the Accomplishment Instructions in Beech Service Bulletin No. 2399, dated March 1991. Actuators must not be removed to paint the .50 by 1 inch stripe on the housing.

NOTE: A left-hand trim tab actuator will have threads on its actuator screw that will rotate clockwise when screwed into the actuator assembly, and a right-hand trim tab actuator will have threads on its actuator screw that will rotate counterclockwise when screwed into the actuator assembly.

(d) Special flight permits may be issued in accordance with FAR 21.197 and 21.199 to operate airplanes to a location where the requirements of this AD can be accomplished.

(e) An alternative method of compliance or adjustment of the compliance time that provides an equivalent level of safety may be approved by the Manager, Wichita Aircraft Certification Office, 1801 Airport Road, Room 100, Mid-Continent Airport, Wichita, Kansas 67209. The request should be forwarded through an appropriate FAA Maintenance Inspector, who may add comments and then send it to the Manager, Wichita Aircraft Certification Office.

(f) The modification required by this AD shall be done in accordance with Beech Service Bulletin No. 2399, dated March 1991. This incorporation by reference was approved by the Director of the Federal Register in accordance with 5 U.S.C. 552(a) and 1 CFR Part 51. Copies may be obtained from the Beech Aircraft Corporation, Commercial Service, Department 52, P.O. Box 85, Wichita, Kansas 67201-0085. Copies may be inspected at the FAA, Central Region, Office of the Assistant Chief Counsel, Room 1558, 601 E. 12th Street, Kansas City, Missouri, or at the Office of the Federal Register, 1100 L Street, NW, Room 8401, Washington, DC.

This amendment (39-7099, AD 91-17-01) becomes effective on November 25, 1991.

**FOR FURTHER INFORMATION CONTACT:**

Mr. Larry Engler, Aerospace Engineer, FAA, Wichita Aircraft Certification Office, 1801 Airport Road, Room 100, Mid-Continent Airport, Wichita, Kansas 67209; Telephone (316) 946-4122.