

Sektion 2. Utlandstillverkad flygmateriel

TITEL: Kontroll av bultar i huvudrotorväxel

GÄLLER: Alla MDHC modell 369D, 369E och 369F/FF

ÅTGÄRD: För att undvika fel i huvudrotorväxelns drivenhet, med haveri som följd, utför åtgärder i enlighet med MDHC Service Information Notice (SIN) DN-166.1, EN-57.1 och SIN FN-45.1 daterade 14 mars 1990 eller senare revisioner

TID FÖR ÅTGÄRD: I enlighet med tider angivna i bifogad kopia av FAA AD 90-24-07 räknat från detta LVD's utgivningsdatum.

UNDERLAG: MDHC Service Information Notice (SIN) DN-166.1, EN-57.1 och SIN FN-45.1 daterade 14 mars 1990 eller senare revisioner. FAA AD 90-24-07 eller senare revision.

REFERENS: FAA AD 90-24-07

BESLUTS-DATUM: 1991-01-24

LFS: 1991:4

Åtgärd enligt LVD utgör nödvändig förutsättning för ifrågavarande flygmateriels luftvärdighet. Referens BCL M 1.11. Anteckning om åtgärd, som vidtagits i enlighet med LVD, skall införas i teknisk journal för berörd flygmateriel med hänvisning till ifrågavarande LVD-nummer. Angivet underlag refererar till senaste gällande revision/utgåva. LVD utges i luftfartsverkets författningssamlingar LFS.



U.S. Department
of Transportation

**Federal Aviation
Administration**

AIRWORTHINESS DIRECTIVE

AVIATION STANDARDS NATIONAL FIELD OFFICE
P.O. BOX 26460
OKLAHOMA CITY, OKLAHOMA 73125

The following Airworthiness Directive issued by the Federal Aviation Administration in accordance with the provisions of Federal Aviation Regulations, Part 39, applies to an aircraft model of which our records indicate you may be the registered owner. Airworthiness Directives affect aviation safety. They are regulations which require immediate attention. You are cautioned that no person may operate an aircraft to which an Airworthiness Directive applies, except in accordance with the requirements of the Airworthiness Directive (FAR 39.3).

90-24-07 MCDONNELL DOUGLAS HELICOPTER COMPANY (MDHC):

Amendment 39-6784. Docket Number 90-ASW-10.

Applicability: All MDHC Model 369D, 369E, and 369F/FF series helicopters certificated in any category.

Compliance: Required as indicated, unless previously accomplished.

To prevent possible failure of the main rotor transmission drive assembly, which could result in loss of control of the helicopter, accomplish the following:

(a) Within the next 300 hours' time in service after the effective date of the AD or at the next annual inspection or the next time the transmission is removed, whichever occurs first, after the main rotor transmission is removed inspect the MS21250-04036 bolts which retain the debris cover, P/N 369D25174. Remove any bolts with the head inscription shown as unacceptable in Figure 1, and replace with MS21250-04038 bolts, which have a length of 2.887 plus/minus 0.010 inch.

NOTE: MDHC Service Information Notice (SIN) DN-166.1, EN-57.1, and SIN FN-45.1, dated March 14, 1990, or later revisions pertain to this subject.

(b) Inspect the thread protrusion of all bolts. Remove any bolt which does not protrude through the H14-4 nut for a length equivalent to two full threads (0.071 inch minimum), including the chamfer. Replace removed bolts with MS21250-04038 bolts. Torque the bolts to 50-70 inch pounds. Verify that the bolts protrude through the nut for a length equivalent to two full threads (0.071 inch minimum), including the chamfer. If more than four threads protrude through the nut, add AN960C416L or AN960C416 washers under the nut as required. Remove and reinstall parts in accordance with the manufacturer's instructions.

(c) Apply a white dot to the main transmission data plate to indicate that the transmission has been inspected and reworked in accordance with the manufacturer's instructions, and record compliance with this AD in the rotorcraft log book.

(d) In accordance with FAR Sections 21.197 and 21.199, flight is permitted to a base where the requirements of this AD may be accomplished.

(e) An alternate method of compliance or adjustment of the compliance time which provides an equivalent level of safety, may be used when approved by the Manager, Los Angeles Aircraft Certification Office, ANM-100L, FAA, Northwest Mountain Region, 3229 East Spring Street, Long Beach, California 90806-2425.

2 90-24-07

This amendment (39-6784, AD 90-24-07) becomes effective on December 10, 1990.

FOR FURTHER INFORMATION CONTACT:

Mr. Roy McKinnon, Aerospace Engineer, ANM-143L, Northwest Mountain Region, Los Angeles Aircraft Certification Office, 3229 East Spring Street, Long Beach, California 90806-2425, telephone (213) 988-5247.

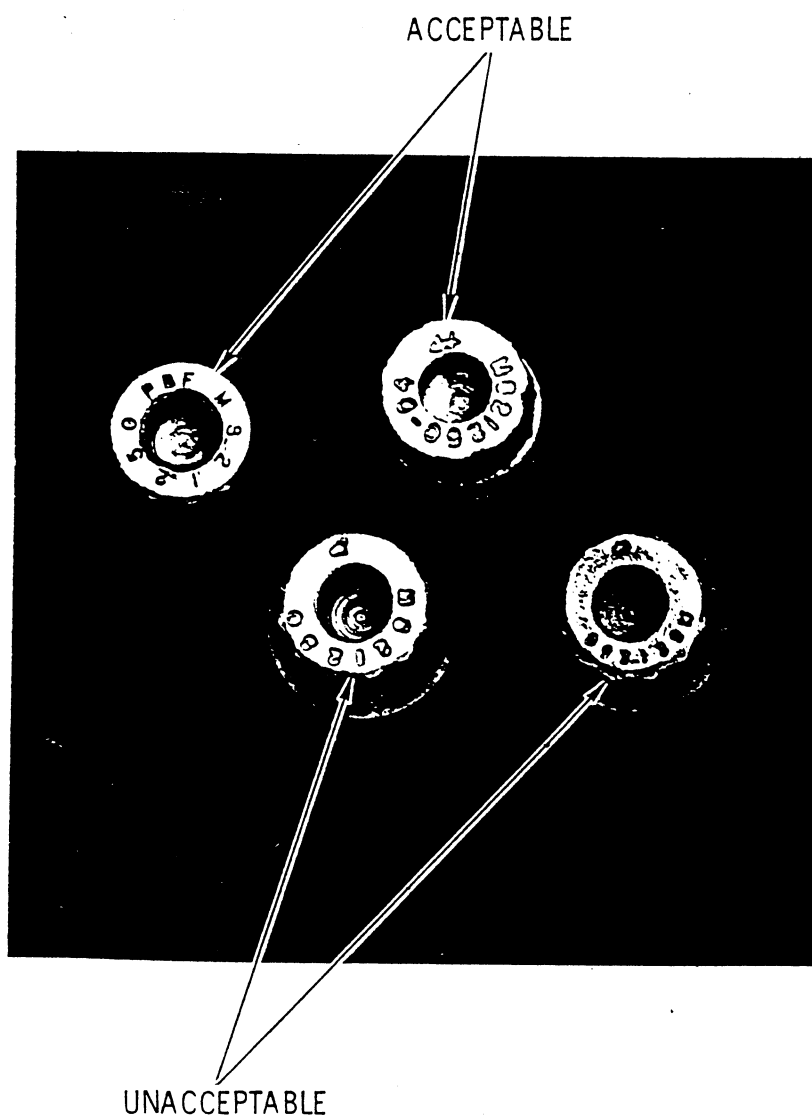


Figure 1. Inspection/Definition of Bolt Heads.