

Sektion 2. Utlandstillverkad flygmateriel

**TITEL:** Kontroll eventuellt byte av hävarm för klaffsystem

**GÄLLER:** Cessna modell 208 och 208A, S/N 20800001 t o m 20800173  
208B S/N 208B0001 t o m 208B0202.

**ÅTGÄRD:** För att undvika felutfall av hävarmsenhet P/N 2622083-18, vilket kan resultera i oönskad infällning av klaffar, utför åtgärder enligt bifogad kopia av FAA AD 90-05-01.

**TID FÖR  
ÅTGÄRD:** Inom 50 flygtimmar från detta LVD:s utgivningsdatum om ej tidigare utförts.

**UNDERLAG:** Cessna Service Bulletin CAB89-34.  
FAA AD 90-05-01

**REFERENS:** FAA AD 90-05-01.

**UTGIVNINGS-  
DATUM:** 1990-04-26

**LFS: 1990:13**

Åtgärd enligt LVD utgör nödvändig förutsättning för ifrågavarende flygmateriels luftvärdighet. Referens BCL M 1.11. Anteckning om åtgärd, som vidtagits i enlighet med LVD, skall införas i teknisk journal för berörd flygmateriel med hänvisning till ifrågavarande LVD-nummer. Angivet underlag refererar till senaste gällande revision/utgåva. LVD utges i luftfartsverkets författningssamlingar LFS.



U.S. Department  
of Transportation  
**Federal Aviation  
Administration**

## AIRWORTHINESS DIRECTIVE

AVIATION STANDARDS NATIONAL FIELD OFFICE  
P.O. BOX 26460  
OKLAHOMA CITY, OKLAHOMA 73125

The following Airworthiness Directive issued by the Federal Aviation Administration in accordance with the provisions of Federal Aviation Regulations, Part 39, applies to an aircraft model of which our records indicate you may be the registered owner. Airworthiness Directives affect aviation safety. They are regulations which require immediate attention. You are cautioned that no person may operate an aircraft to which an Airworthiness Directive applies, except in accordance with the requirements of the Airworthiness Directive (FAR 39.3).

90-05-01 CESSNA: Amendment 39-6513.

**Applicability:** Model 208 and 208A (Serial Numbers 20800001 thru 20800173), and 208B (Serial Numbers 208B0001 through 208B0202) airplanes certificated in any category.

**Compliance:** Required within the next 50 hours time-in-service (TIS) after the effective date of this AD, unless already accomplished.

To prevent failure of the Part Number (P/N) 2622083-18 bellcrank assembly and the resulting unexpected retraction of the flaps, accomplish the following:

(a) Visually inspect the P/N 2622083-18 bellcrank assembly for cracks, deformation, and incomplete welds in accordance with Cessna Service Bulletin CAB89-34, dated December 22, 1989.

(1) If any cracks or deformation are found, prior to further flight replace the P/N 2622083-18 bellcrank with an airworthy bellcrank.

(2) If any incomplete welds are found, within the next 50 hours TIS replace the P/N 2622083-18 bellcrank with an airworthy bellcrank.

(3) If any bellcranks are to be replaced, inspect the replacement part in accordance with Cessna SB CAB89-34 dated December 22, 1989 before installation.

(b) Airplanes may be flown in accordance with FAR 21.197 to a location where this AD may be accomplished.

(c) An alternate method of compliance or adjustment of the initial or repetitive compliance times which provides an equivalent level of safety may be approved by the Manager, Wichita Aircraft Certification Office, 1801 Airport Road, Room 100, Wichita, Kansas 67209.

**NOTE:** The request should be forwarded through an FAA Maintenance Inspector, who may add comments and then send it to the Manager, Wichita Aircraft Certification Office.

All persons affected by this directive may obtain copies of the document referred to herein upon request to the Cessna Aircraft Company, Customer Service, P.O. Box 7704, Wichita, Kansas 67217, or may examine this document at the FAA, Central Region, Office of the Assistant Chief Counsel, Room 1558, 601 East 12th Street, Kansas City, Missouri 64106.

This amendment (39-6513, AD 90-05-01) becomes effective on March 5, 1990.

**FOR FURTHER INFORMATION CONTACT:** Douglas W. Haig, Aerospace Engineer, Federal Aviation Administration, Wichita Aircraft Certification Office, 1801 Airport Road, Room 100, Wichita, Kansas 67209; Telephone 316-946-4409.