LUFTFARTSVERKET 601 79 NORRKÖPING, SVERIGE TEL: 011 - 19 20 00 TELEGRAMADRESSER: LUFTFARTSNÄTET ESKLYA KOMM CIVILAIR NORRKÖPING TELEX 64250

NOTE:

(Cont)

AIRWORTHINESS DIRECTIVES

LUFTVÄRDIGHETS-ANVISNINGAR (LVA)

Aircraft

SAAB - 2E

Cancels and substitutes

SAAB - 2D



AD NO.

444 Subject: Reinforcement of the rudder trim weight SAAB SAFIR 91A, 91B, 91B-2, 91C and 91D Applies to: One case of rudder locking due to loose trim weight has occurred Reason: Dismount and carry out reinforcement in accordance with SAAB Modification Bulletin Required action: No 91.3/5, dated 3.4.70 Date: 1970-04-20 Within 50 flying hours from the effective date of this Airworthiness Directive, unless compliance: already accomplished 1519A Subject: Replacement of bolts and inspection for cracks Applies. SAAB SAFIR 91B, 91B2, 91C and 91D Possibility of fatigue failure in specific joints on aircraft with a considérable amount Reason: of flying hours Required action: Replace wing coupling bolts and bolts connecting engine mount to fuselage and bolts connecting engine to engine mount, and check the fin and stabilizer front fitting for cracks and carry out an ultrasonic test or a dye penetrant inspection, in and around the holes for the bolts, attaching the lower brackets to the wing and fuselage spar, in accordance with SAAB-SCANIA Service Bulletin No. 91.1.003, Issue 2, dated March, 1979 SAF 1980-11-03 Time of Wing coupling bolts must be replaced on aircraft with more than 3000 hours in service compliance: but before 5000 hours in service. Thereafter at intervals not to exceed 3000 hours in service. 2 Bolts connecting engine mount to fuselage and bolts connecting engine to engine mount must be replaced every 5000 hours in service 3 Fin and stabilizer front fitting must be checked for cracks and if necessary exchanged at 5000 hours in service The area in and around the holes for the bolts attaching the lower brackets to the wing and the fuselage spar must be ultrasonic or dye penetrant checked at 6000 hours in service If the aircraft after 6000 hours in service shall be used for aerobatic, $\max N_{\sigma} = 4.8$ may be allowed. Described inspection (4) shall than be performed every 350 hours in If the aircraft after 6000 hours in service will be used according to FAR 23 category, Normal or Utility, the inspection (4) is necessary only every 1000 hours in service. 1685 Dye penetrant inspection of components in flight control system Subject: SAAB SAFIR 91B, 91B2, 91C, and 91D, serial numbers S/N 91,201 through 91,474 and spare Applies to: parts Reason: Cracks have been found in bellcranks manufactured from Hydronaulium Required action: Inspect certain components in the flight control system by using dye penetrant method in accordance with SAAB-SCANIA Service Bulletin No 91.4.002, dated November, 1980. Cracked components must be replaced. 2 Acceptable components must be heat treated prior to re-installation in accordance with the same Service Bulletin. 3 Report findings of cracks to SAAB-SCANIA AB, Linkoeping, Sweden.

1982-09-27

application of the flight controls should be avoided.

Until action above is accomplished, all aerobatic maneuvres are prohibited and abrupt

AD NO. SAAB - 2E (Cont) Date: 1980-12-01 Time of Within 50 flight hours from the effective date of this Airworthiness Directive, compliance: or latest May 1, 1981, whichever occurs first. Inspection of elevator spar 1832 Subject: Applies to: SAAB-SAFIR 91B, 91B2, 91C, and 91D Damage to elevator spar at bearing assembly attachment has been reported in a few cases Reason: Carry out the required action in accordance with SAAB-SCANIA Service Bulletin No 91.1.004, Issue 2 $\,$ Required action: 1982-09-27 Date: Time of compliance: Before further flight, if not already performed. Thereafter at intervals not exceeding 200 flight hours.