
Sektion 2. Utlandstillverkad flygmateriel

TITEL: Sprickkontroll av stjärtrotorbladen.

GÄLLER: Eurocopter modeller AS 350B, BA, B1, B2, B3, BB och D, AS 355E, F, F1, F2 och N utrustade med stjärtrotorblad P/N enligt Eurocopter Alert Telex 05.00.40 och 05.00.38.

ÅTGÄRD: Utför åtgärder enligt bifogad kopia av DGAC TAD T2001-640-089(A) och T2001-641-067(A).

TID FÖR ÅTGÄRD: Enligt DGAC TAD T2001-640-089(A) och T2001-641-067(A).

UNDERLAG: Eurocopter AS 350 Alert Telex No 05.00.40 och AS 355 Alert Telex No 05.00.38

REFERENS: Telegraphic AD T2001-640-089(A), T2001-641-067(A)

BESLUTSDATUM: 9 januari 2002

LFS 2002:06

Åtgärder enligt LVD utgör nödvändig förutsättning för ifrågavarande flygmateriels luftvärdighet. Referens BCL M 1.11.
Anteckning om åtgärd, som vidtagits i enlighet med LVD, skall införas i teknisk journal för berörd flygmateriel med hänvisning till ifrågavarande LVD-nummer. Angivet underlag refererar till senast gällande revision/utgåva. LVD utges i luftfartsverkets författningssamlingar LFS.

Postadress	Gatuadress	Telefonnummer	Faxnummer	Telegram Civilair	Telex
601 79 NORRKÖPING	Vikobplan 11	011-19 20 00	011-19 25 75	Norrköping	62450

01 12/20 15:43 FAX 4611192575

LFV NORRKÖPING
GSAC: 0146904840

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Page 002/002

TRANSLATION (in case of difficulty, refer to french text)

TELEGRAPHIC AIRWORTHINESS DIRECTIVE (TAD) RELEASED BY DIRECTION GENERALE DE L'AVIATION CIVILE (FRANCE).

THIS TEXT IS NOT SENT TO FOREIGN OPERATORS OF AIRCRAFT NON REGISTERED IN FRANCE. IT IS UP TO THE NATIONAL AUTHORITIES TO FORWARD IT TO THEM ON RECEIPT.

No. T2001-640-089(A) - EUROCOPTER AS 350 Helicopters

No. T2001-641-067(A) - EUROCOPTER AS 355 Helicopters

SUBJECT : Tail Rotor Blades (ATA 05, 64)

1. EFFECTIVITY:

1.1 AS 350 helicopters, versions B, BA, B1, B2, B3, BB and D

equipped with tail rotor blades P/Nos.:

- 355A12-0031-01/ -02/ -03/ -04/ -05/ -06/ -07/ -08/ -09/ -11/ -12/ -13/ -14,
- 355A12-0040-00/ -01/ -02/ -03/ -04/ -05/ -07/ -08,
- 355A12-0050-04.

1.2 AS 355 helicopters, versions E, F, F1, F2 and N

equipped with tail rotor blades P/Nos.:

- 355A12-0031-01/ -02/ -03/ -04/ -05/ -06/ -07/ -08/ -09/ -11/ -12/ -13/ -14,
- 355A12-0040-00/ -01/ -02/ -03/ -04/ -05/ -07/ -08,
- 355A12-0050-00/ -01/ -02/ -03/ -04/ -05.

2. REASON:

This Airworthiness Directive (AD) is issued following a report of separation of a tail rotor blade trailing edge section due to crack growth in the blade skin. The unbalance caused by the loss of a blade section can cause the TGB to be torn off the tail boom.

3. MANDATORY ACTIONS AND COMPLIANCE TIME:

3.1. From the effective date of this AD, at each check after the last flight of the day, without exceeding 10 flight hours between each check, visually check the area A of the blade pressure face and blade suction face of the tail rotor blades listed in paragraph 1 above, for cracks, in compliance with the instructions given in paragraph 2.A of the referenced Alert Telex.

3.2. If a crack is found in the skin, before the next flight, remove and replace the damaged blade in compliance with the instructions given in paragraph 2.B of the referenced Alert Telex.

NOTE: This check can be carried out by an appropriately trained pilot.

4. REFERENCES: EUROCOPTER AS 350 Alert Telex No.05.00.40
AS 355 Alert Telex No 05.00.38

5. EFFECTIVE DATE: On receipt.

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