



LUFTFARTSVERKET
Luftfartsinspektionen

LUFTVÄRDIGHETSDIREKTIV (LVD)

A Motordrivna Luftfartyg
Socata
LVD Nr 2-3155

Sektion 2. Utlandstillverkad flygmateriel

TITEL: Kontroll/byte av vingens infästningsbultar

GÄLLER: TB9, TB10, TB200, TB20 och TB21, S/N angivna i bifogad kopia av DGAC AD 2001-306(A) med detaljer levererade t.o.m. 15 februari 2001 installerade eller ej i luftfartyg

ÅTGÄRD: Utför åtgärder angivna i DGAC AD 2001-306(A)

TID FÖR ÅTGÄRD: Vid nästa planerade 100 timmarstillsyn men ej senare än 31 december 2001

UNDERLAG: SB SOCATA TB No. BS 10-123-57 daterad juni 2001 eller senare utgåva

REFERENS: DGAC AD 2001-306(A)

BESLUTSDATUM: 28 augusti 2001

LFS 2001:108

Åtgärder enligt LVD utgör nödvändig förutsättning för ifrågavarande flygmateriels luftvärdighet. Referens BCL M 1.11.
Anteckning om åtgärd, som vidtagits i enlighet med LVD, skall införas i teknisk journal för berörd flygmateriel med hänvisning till ifrågavarande LVD-nummer. Angivet underlag refererar till senast gällande revision/utgåva. LVD utges i luftfartsverkets författningssamlingar LFS.

Postadress	Gatuadress	Telefonnummer	Faxnummer	Telegram Civilair	Telex
601 79 NORRKÖPING	Vikobplan 11	011-19 20 00	011-19 25 75	Norrköping	62450

GSAC

AIRWORTHINESS DIRECTIVE

released by DIRECTION GENERALE DE L'AVIATION CIVILE

Inspection and/or modifications described below are mandatory. No person may operate a product to which this Airworthiness Directive applies except in accordance with the requirements of this Airworthiness Directive.

Translation of 'Consigne de Navigabilité' ref. : 2001-306(A)
In case of any difficulty, reference should be made to the French original issue.

SOCATA

TB 9, TB 10, TB 200, TB 20 and TB 21 aircraft

Wings front attaching bolts (ATA 57)

1. APPLICABILITY:

The present Airworthiness Directive is applicable to all SOCATA TB 9, TB 10, TB 200, TB 20 and TB 21 aircraft S/N 2040, 2043, 2044, 2049 through 2053, 2056, 2057, 2064, 2069 through 2071 and the axes supplied as spare parts by SOCATA, from the February 15, 2001 installed or not on aircraft.

2. REASON:

A batch of wing front attachment screws used on TB's aircraft and supplied as spare parts present an improper threading.

3. MANDATORY ACTIONS AND COMPLIANCE TIME:

The following measures are made mandatory from the effective date of this Airworthiness Directive:

At the next 100 hours maintenance scheduled visit and before December 31, 2001, disassemble and replace the faulty axes according to the instructions of SOCATA TB Service Bulletin No. BS 10-123-57.

Record the application of this Airworthiness Directive in the aircraft log book.

REF.: SB SOCATA TB No. BS 10-123-57 dated June 2001
(or later revisions).

EFFECTIVE DATE : AUGUST 04, 2001

n/RH

July 25, 2001

SOCATA
TB 9, TB 10, TB 200, TB 20 and TB 21 aircraft

2001-306(A)