
Sektion 2. Utlandstillverkad flygmateriel

TITEL: Korrosion/sprickkontroll av landställscyliner

GÄLLER: Modell FA-200

ÅTGÄRD: Utför åtgärder angivna i bifogad kopia av TCD-5230-2000

TID FÖR ÅTGÄRD: Inom tider och intervall angivet under Compliance räknat från detta LVD's beslutsdatum

UNDERLAG: Fuji Heavy Industry Service Bulletin No. 200-013 daterad 28 september 1999 eller senare utgåva

REFERENS: TCD-5320-2000

BESLUTSDATUM: 2000-05-02

LFS 2000:59

Åtgärder enligt LVD utgör nödvändig förutsättning för ifrågavarande flygmateriels luftvärdighet. Referens BCL M 1.11.
Anteckning om åtgärd, som vidtagits i enlighet med LVD, skall införas i teknisk journal för berörd flygmateriel med hänvisning till ifrågavarande LVD-nummer. Angivet underlag refererar till senast gällande revision/utgåva. LVD utges i luftfartsverkets författningssamlingar LFS.

Postadress	Gatuadress	Telefonnummer	Faxnummer	Telegram	Telex
601 79 NORRKÖPING	Vikobplan 11	011-19 20 00	011-19 25 75	Civilair Norrköping	62450

ICN

TCF-50-001-1E-1

KU-KI-234	No. TCD-5230-2000
Date <u>March 30, 2000</u> Japan Civil Aviation Bureau TAIKUSEI-KAIZEN-TSUHO <u>Airworthiness Directive</u> The undermentioned examinations or modifications are mandatory.	
1. Applies to : Fuji Model FA-200 series airplanes	
2. Compliance required as indicated, unless already accomplished. To prevent corrosion on the internal surface of landing gear cylinder assembly, accomplish the following. 2.1 Within 100 flight hours after the effective date of this AD, and thereafter within intervals not to exceed 100 flight hours from the previous inspection, inspect the external surface of landing gear cylinder for crack using 5X to 10X magnifying glass, in accordance with the paragraph 6(2) of Fuji Heavy Industry Service Bulletin No.200-013 dated September 28, 1999 or JCAB approved later revisions (SB). If crack is found, replace the cylinder with a serviceable part prior to further flight. 2.2 Within one year from the effective date of this AD, and thereafter within intervals not to exceed 1,000 flight hours from the previous inspection, inspect the internal surface of landing gear cylinder assembly for corrosion in accordance with the paragraph 12 of SB. If corrosion is found, remove the corrosion, measure wall thickness of the cylinder and , if necessary, replace the cylinder with a serviceable part prior to further flight. 2.3 First inspection of the paragraph 2.2 of this AD constitutes the	

TCF-50-001-2E

Japan Civil Aviation Bureau	No. TCD-5230-2000	sheet 2/2
terminating action of repetitive inspection of the paragraph 2.1. 2.4 Applying of epoxy polyamide primer in accordance with the paragraph 12-1(9) or 12-2(9) of SB constitute the terminating action of repetitive inspection of the paragraph 2.2 of this AD. 2.5 An alternative means of compliance with this AD may be used, if approved by the Director-General of JCAB.		
3. Remarks 3.1 This AD becomes effective on April 20, 2000. 3.2 Fuji Heavy Industry Service Bulletin No.200-013 and JCAB-approved later revisions pertain to this subject.		