

LUFTVÄRDIGHETSDIREKTIV (LVD)

A. Helikopter Bell (BHTC) LVD Nr 2907

Sektion 2. Utlandstillverkad flygmateriel

TITEL:

Sprickkontroll av stjärtbom

GÄLLER:

Modell 206L enligt bifogad kopia av CF-98-27

ÅTGÄRD:

Utför åtgärder angivna i CF-98-27

TID FÖR ÅTGÄRD:

Efter 18 oktober 1998 inom tider och intervaller angivna i CF-98-27 om ej

tidigare utfört

UNDERLAG:

CF-98-27 och där angivet underlag

REFERENS:

CF-98-27

BESLUTSDATUM:

1998-09-28

LFS 1998:62

Åtgärder enligt LVD utgör nödvändig förutsättning för ifrågavarande flygmateriels luftvärdighet. Referens BCL M 1.11.

Anteckning om åtgärd, som vidtagits i enlighet med LVD, skall införas i teknisk journal för berörd flygmateriel med hänvisning till ifrågavarande LVD-nummer. Angivet underlag refererar till senast gällande revision/utgåva. LVD utges i luftfartsverkets författningssamlingar LFS.

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Gatuadress

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Bilaga till LVD 2907



Aviation civile

TP 7245E No. 1/2 CF-98-27 Date 31 August 1998

AIRWORTHINESS DIRECTIVE

THE FOLLOWING AIRWORTHINESS DIRECTIVE (AD) MAY BE APPLICABLE TO AN AIRCRAFT, WHICH OUR RECORDS INDICATE IS REGISTERED IN YOUR NAME. ADS ARE ISSUED PURSUANT TO CARADIAN AVIATION REGULATION (CAR) 593. PURSUANT TO CAR 605.84 AND THE FURTHER DETAILS OF CAR STANDARD 625, APPENDIX H, THE CONTINUING AIRWORTHINESS OF AN AIRCRAFT IS CONTINGENT UPON COMPLIANCE WITH ALL APPLICABLE ADS. FAILURE TO COMPLY WITH THE REQUIREMENTS OF AN AD MAY INVALIDATE THE FLIGHT AUTHORIZATION OF THE AIRCRAFT. ALTERNATIVE MEANS OF COMPLIANCE SHALL BE APPLIED FOR IN ACCORDANCE WITH CAR 605.84 AND THE ABOVE-REFERENCED STANDARD.

CF- 98-27 BELL

Applies to Bell Helicopter Textron Canada (BHTC) Model 206L series helicopters as indicated. Tailboom assemblies modified in accordance with Technical Bulletin (TB) 206L-96-191, or BHT-206-SRM-1, section 6-3-4, with upper tailboom skin P/N 206-033-004-155 installed, are not affected by this directive.

Compliance is required as indicated, unless already accomplished.

Cracking has been reported on tailbooms of earlier designs. If left undetected, these cracks could lead to failure of the tailboom and loss of control of the helicopter. Bell has issued Alert Service Bulletins (ASB) 206L-87-47 Revision C, dated 23 October 1989, and 206L-97-107 Revision A, dated 4 February 1998, detailing inspections required to detect cracks and providing applicable modifications.

To ensure the structural integrity of all affected tailbooms, accomplish the following measures as applicable:

Part 1.

Applies to Model 206L series helicopters, S/N 45001 through 51283, having unmodified tailboom assemblies P/N 206-033-004-003, -011, -045 and -103.

Within the next 200 hours air time after the effective date of this directive, but not later than 31 May 1999, modify the tailboom between boom station BS 151.15 and 159.35 by installing doubler P/N 206-704-167-111 with clip P/N 206-033-407-119, and re-identify the tailboom by adding the suffix "FM" to the original part number as outlined in Part I of ASB 206L-87-47 Revision C.

Part 2.

Applies to the following Model 206L series helicopters:

Model 206L S/N 45003 through 45153, and 46601 through 46617

Model 206L-1 S/N 45154 through 45790 Model 206L-3 S/N 51001 through 51283

having tailboom assemblies of the following part numbers:

206-033-004-003FM, -003FM1 206-033-004-011FM, -011FM1, -011FM2, -011FM3 206-033-004-103FM, -103FM1 206-033-004-045FM



No.	CF-98-27	2/2
TIA.		

- (a) For tailbooms with a sheet metal tail rotor gearbox support, within the next 100 hours air time after the effective date of this directive and at every $1000 \ (\pm 10\%)$ hours air time thereafter, perform an Eddy Current inspection as outlined in Part I of ASB 206L-97-107 Revision A.
- (b) For tailbooms with a cast tail rotor gearbox support P/N 206-033-426-001 and that do not have upper skin P/N 206-033-004-155 installed, within the next 200 hours air time after the effective date of this directive, but not later than 31 May 1999, install nutplate assemblies as per Part II of ASB 206L-97-107 Revision A.

Part 3.

For all Model 206L helicopters affected by Parts 1 and 2 above, replacement of the tailboom with any tailboom listed below constitutes terminating action for the requirements of Parts 1 and 2 of this directive:

206-033-004-003FM2 206-033-004-011FM4 206-033-004-103FM2 206-033-004-045FM1, -045FM2 206-033-004-143, -143FM1 206-033-004-175, -175FM1 206-033-004-177

Note: The reference to MIL-STD-6866, for fluorescent dye penetrant inspection as listed in ASB 206L-97-107 Revision A, Part II, paragraph 4, page 9, has since been replaced by American Society for Testing and Materials Standard ASTM E1417-95a, "Practice for Liquid Penetrant Examination".

This directive become effective 18 October 1998

For Minister of Transport

J.E. Hurley

Acting Chief, Continuing Airworthiness

For further information contact a Transport Canada Centre, or Mr. Bogdan Gajewski, Continuing Airworthiness, Ottawa, telephone (613) 952-4450, facsimile (613) 996-9178 or e-mail gajewsb@tc.gc.ca.