
Sektion 2. Utlandstillverkad flygmateriel

TITEL: Sprickkontroll av stjärtrotorblad

GÄLLER: SA 313B, SE 3130, SA 3180, SA 318B och SA 318C med installerade stjärtrotorblad P/N 3160S-34-10000 alla strecknummer eller P/N 3160S-34-11000 alla strecknummer

ÅTGÄRD: Utför åtgärder angivna i bifogad kopia av DGAC AD 1998-286-056(A) R1

TID FÖR ÅTGÄRD: Inom 10 flygtimmar därefter i intervall av 10 flygtimmar

UNDERLAG: EUROCOPTER ALOUETTE II Alert Service Bulletin n° 05-94

REFERENS: DGAC 1998-286-056(A) R1

BESLUTSDATUM: 2 april 2001

LFS 2001:50

Åtgärder enligt LVD utgör nödvändig förutsättning för ifrågavarande flygmateriels luftvärdighet. Referens BCL M 1.11.
Anteckning om åtgärd, som vidtagits i enlighet med LVD, skall införas i teknisk journal för berörd flygmateriel med hänvisning till ifrågavarande LVD-nummer. Angivet underlag refererar till senast gällande revision/utgåva. LVD utges i luftfartsverkets författningssamlingar LFS.

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GSAC

AIRWORTHINESS DIRECTIVE

released by DIRECTION GENERALE DE L'AVIATION CIVILE

Inspection and/or modifications described below are mandatory. No person may operate a product to which this Airworthiness Directive applies except in accordance with the requirements of this Airworthiness Directive.

Translation of 'Consigne de Navigabilité' ref. : 1998-286-056(A) R1
In case of any difficulty, reference should be made to the French original issue.

EUROCOPTER

SA 313/318 helicopters

Tail rotor blades - Skin (ATA 64)

1. APPLICABILITY

SA 313B, SE 3130, SA 3180, SA 318B and SA 318C, helicopters equipped with tail rotor blades P/N :

- 3160S-34-10 000 all dash numbers
- 3160S-34-11 000 all dash numbers.

2. REASONS

This AD is prompted subsequently to the discovery of a crack in the skin of a blade lower surface, near to the attachment bolt on the blade cuff stem, on the leading edge side of a tail rotor blade, which can lead to the loss of the blade.

Revision 1 of this Airworthiness Directive takes into account the conversion of Service Telex No. 05.94 into Alert Service Bulletin (ASB) No. 05.94.

3. COMPLIANCE TIME

The actions stipulated in paragraph 4 were to be taken a first time within the 10 flight hours following the effective date of the original telegraphic Airworthiness Directive, then every 10 flight hours.

These actions are still to be taken every 10 flight hours.

4. MANDATORY ACTIONS

4.1. On the lower and upper surfaces of blades P/Nos:

3160S-34-10000-10 and onwards,
3160S-34-11000 all dash numbers,

perform a visual crack check in compliance with the operational procedure defined in paragraph 2.B of the referenced ASB.

4.2. If a crack is found, remove the blade before next flight.

Reminder: Blades P/Nos. 3160S-34-11000-08 and 09 must be checked before each flight in compliance with Maintenance Manual chapter 5.1.

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n/DJ

March 21, 2001

EUROCOPTER
SA 313/318 helicopters

1998-286-056(A) R1

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AIRWORTHINESS DIRECTIVE

ref. : 1998-286-056(A) R1

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REF.: EUROCOPTER ALOUETTE II Alert Service Bulletin n° 05-94.

This Revision 1 replaces Airworthiness Directive 98-286-056(A) dated August 12, 1998.

EFFECTIVE DATES :

Original AD : On receipt of telegraphic AD
issued on JULY 13, 1998

Revision 1 : MARCH 31, 2001