

## LUFTVÄRDIGHETSDIREKTIV (LVD)

A. Flygplan Raytheon (Beech) LVD Nr 2860 Upphäver LVD 2728

Sektion 2. Utlandstillverkad flygmateriel

TITEL:

Byte av "Outflow/Safety valve"

**GÄLLER:** 

Modellerna 400, 400A, MU 300 och MU 300-10, S/N angivna i bifogad

kopia av FAA AD 98-06-30

**ATGÄRD:** 

Utför åtgärder angivna i FAA AD 98-06-30

TID FÖR ATGÄRD:

Inom 18 månader räknat från detta LVD's utgivningsdatum

**UNDERLAG:** 

FAA AD 98-06-30

Service Bulletiner angivna i FAA AD 98-06-30

**REFERENS:** 

FAA AD 98-06-30

**BESLUTSDATUM:** 

1998-04-24

LFS 1998:28

Åtgärder enligt LVD utgör nödvändig förutsättning för ifrågavarande flygmateriels luftvärdighet. Referens BCL M 1.11.

Anteckning om åtgärd, som vidtagits i enlighet med LVD, skall införas i teknisk journal för berörd flygmateriel med hänvisning till ifrågavarande LVD-nummer. Angivet underlag refererar till senast gällande revision/utgåva. LVD utges i luftfartsverkets författningssamlingar LFS.

**Postadress** 

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## AIRWORTHINESS DIRECTIVE

REGULATORY SUPPORT DIVISION
P.O. BOX 26460
OKLAHOMA CITY, OKLAHOMA 73125-0460

U.S. Department of Transportation Federal Aviation Administration

The following Airworthiness Directive issued by the Federal Aviation Administration in accordance with the provisions of Federal Aviation Regulations, Part 39, applies to an aircraft model of which our records indicate you may be the registered owner. Airworthiness Directives affect aviation safety and are regulations which require immediate attention. You are cautioned that no person may operate an aircraft to which an Airworthiness Directive applies, except in accordance with the requirements of the Airworthiness Directive (reference FAR Subpert 39.3).

98-06-30 RAYTHEON AIRCRAFT COMPANY (Formerly Beech, Raytheon Corporate Jets, British Aerospace, Hawker Siddley, et al.): Amendment 39-10408. Docket 97-NM-68-AD. Supersedes AD 96-17-10, Amendment 39-9719.

Applicability: The following models and series of airplanes, certificated in any category, equipped with AlliedSignal outflow/safety valves, as identified in AlliedSignal Aerospace Service Bulletin 103570-21-4012, Revision 1, dated May 30, 1995:

Model of AirplaneSerial Numbers400RJ-1 through RJ-65 inclusive400ARK-1 through RK-42 inclusive400T (military)TT-4 and TT-19MU-300S/N A001SA through A091SA inclusiveMU-300-10A1001SA through A1011SA inclusive

NOTE 1: This AD applies to each airplane identified in the preceding applicability provision, regardless of whether it has been modified, altered, or repaired in the area subject to the requirements of this AD. For airplanes that have been modified, altered, or repaired so that the performance of the requirements of this AD is affected, the owner/operator must request approval for an alternative method of compliance in accordance with paragraph (c) of this AD. The request should include an assessment of the effect of the modification, alteration, or repair on the unsafe condition addressed by this AD; and, if the unsafe condition has not been eliminated, the request should include specific proposed actions to address it.

Compliance: Required as indicated, unless accomplished previously.

To prevent cracking and consequent failure of the outflow/safety valves, which could result in rapid decompression of the airplane, accomplish the following:

- (a) Replace the outflow/safety valve in accordance with AlliedSignal Aerospace Service Bulletin 103570-21-4012, Revision 1, dated May 30, 1995, or Raytheon Service Bulletin No. 2476, Revision II, dated June 1997, at the time specified in paragraph (a)(1) or (a)(2) of this AD, as applicable.
- (1) For Model 400, 400A, MU-300-10 series airplanes: Replace within 18 months after September 24, 1996 (the effective date of AD 96-17-10, amendment 39-9719).
- (2) For Model 400T (military) and MU-300 series airplanes: Replace within 18 months after the effective date of AD.
- (b) As of the effective date of this AD, no person shall install an outflow/safety valve, having a part number and serial number identified in AlliedSignal Aerospace Service Bulletin 103570-21-4012, Revision 1, dated May 30, 1995, on any airplane unless that valve is considered to be serviceable in accordance with the applicable service bulletin.
- (c) An alternative method of compliance or adjustment of the compliance time that provides an acceptable level of safety may be used if approved by the Manager, Wichita Aircraft Certification Office (ACO), FAA, Small Airplane Directorate. Operators shall submit their requests through an appropriate FAA Principal Maintenance Inspector, who may add comments and then send it to the Manager, Wichita ACO.
- NOTE 2: Information concerning the existence of approved alternative methods of compliance with this AD, if any, may be obtained from the Wichita ACO.
- (d) Special flight permits may be issued in accordance with sections 21.197 and 21.199 of the Federal Aviation Regulations (14 CFR 21.197 and 21.199) to operate the airplane to a location where the requirements of this AD can be accomplished.
- (e) The replacement shall be done in accordance with AlliedSignal Aerospace Service Bulletin 103570-21-4012, Revision 1, dated May 30, 1995; or Raytheon Service Bulletin No. 2476, Revision II, dated June 1997.
- (1) The incorporation by reference of Raytheon Service Bulletin No. 2476, Revision II, dated June 1997, is approved by the Director of the Federal Register, in accordance with 5 U.S.C. 552(a) and 1 CFR part 51.
- (2) The incorporation by reference of AlliedSignal Aerospace Service Bulletin 103570-21-4012, Revision 1, dated May 30, 1995, was approved previously by the Director of the Federal Register as of September 24, 1996 (61 FR 42996, August 20, 1996).

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- (3) Copies may be obtained from AlliedSignal Aerospace, Technical Publications, Dept. 65-70, P.O. Box 52170, Phoenix, Arizona 85072-2170; or Raytheon Aircraft Company, Manager Service Engineering, Hawker Customer Support Department, P.O. Box 85, Wichita, Kansas 67201-0085. Copies may be inspected at the FAA, Transport Airplane Directorate, 1601 Lind Avenue, SW., Renton, Washington; or at the FAA, Small Airplane Directorate, Wichita Aircraft Certification Office, 1801 Airport Road, Room 100, Mid-Continent Airport, Wichita, Kansas; or at the Office of the Federal Register, 800 North Capitol Street, NW., suite 700, Washington, DC.
  - (f) This amendment becomes effective on April 24, 1998.

## FOR FURTHER INFORMATION CONTACT:

Michael D. Imbler, Aerospace Engineer, Systems and Propulsion Branch, ACE-116W, FAA, Small Airplane Directorate, Wichita Aircraft Certification Office, 1801 Airport Road, Room 100, Mid-Continent Airport, Wichita, Kansas 67209; telephone (316) 946-4147; fax (316) 946-4407.